

ISO 9001 Certified

Catalog PPD-5209/USA



Aerospace Piston and Rod T-Seals







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Parker Aerospace T-Seals



PARKER's Aerospace T-Seals employ the latest in compounding and design technology to deliver reliable, effective sealing. Parker Aerospace T-Seals

are available for AS4716 / MIL-G-5514F and MS28775 cross-sections and diameters from Dash Size No. -006 through -460. They are available for rods and pistons, and for zero back-up, one back-up and two back-up groove width O-ring replacements.

Parker's Aerospace T-Seals are manufactured under strict procedures conforming to MIL-I-45208. Parker's manufacturing facilities are qualified and certified to the ISO 9001 standard. A wide range of seal and back-up compounds from Parker's advanced materials technology is available to match your application requirements. These seals meet all applicable military specifications.

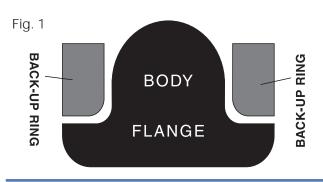


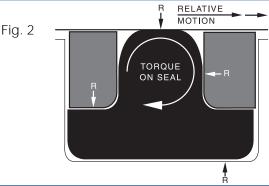
THE T-SEAL DEVELOPMENT

Parker Aerospace T-seals replace existing O-ring seals in long-lived aircraft hydraulic and pneumatic systems. Their ability to fit into existing grooves means that a retrofit can be made with no re-machining or major revisions to the existing hardware design. This simplifies both field changes and new equipment manufacturing.

THE SEALING ELEMENT

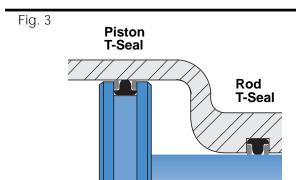
The T-seal's distinctive cross-section is extremely functional. The elastomeric seal element (See Figure 1) consists of a flange and a body, each of which has a multiple purpose. The flange provides the static seal against the bottom of the groove, provides positive radial actuation of the back-up rings, and stabilizes the seal against rolling in the grooves. The body provides the squeeze or interference seal against the dynamic surface, loads the flange to enhance the static seal against the groove, and contributes elastomeric mass to the downstream flange to displace the back-up ring radially. The squeeze effected during installation duplicates that of an O-ring, giving the T-seal its ability to seal at the lowest pressures. Finally, the square or rectangular shape of the T-seal assembly with its back-up rings eliminates the spiral failure mode characteristic of O-rings. Parker T-seals cannot spiral fail (See Figure 2).

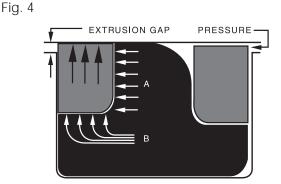






Introduction





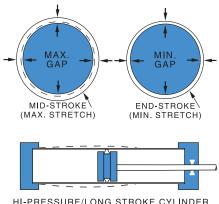
BACK-UP RINGS

Back-up rings are on the outside diameters of piston T-seals, and the inside diameters of rod T-seals (See Figure 3). The unique "hydraulic" loading of the backup rings by the T-seal flange occurs when the installed seal assembly is exposed to differential pressure causing positive back-up ring actuation. Positive actuation allows the T-seal back-up rings to respond more rapidly than the plastic deformation causes Oring back-up rings to close an extrusion gap. For this reason T-seals adapt to wider gaps, and respond to shocks and pressure surges immediately by increasing the radial force against the dynamic surface (See Figure 4).

It should be remembered that extrusion gaps are seldom constant throughout a stroke. The gap may vary due to uneven wear along a rod or bore, it may vary as system pressure rises (breathing) and distance from end restraints changes (thereby changing the stretch of cylinder walls). Ovality of cylinder tubes or rod glands due to side-load is not necessarily constant throughout the stroke, and outof-round tubes may assume a near-perfect circular shape where restrained by the end caps (See Figures 5 & 6).

These numerous possible sources of changing extrusion gaps make Parker T-seals' speed of

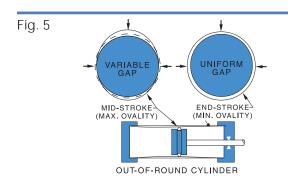
Fig. 6

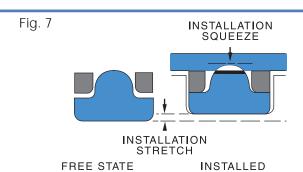


HI-PRESSURE/LONG STROKE CYLINDER

response extremely valuable. Despite rapid strokes, lateral shock loads, pressure surges, and uneven wear or stretching, Parker T-seals maintain zero extrusion gaps.

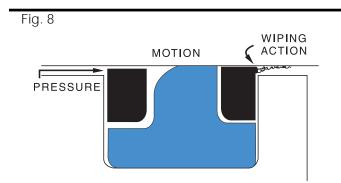
In the free state, back-up rings appear quite loose on their seal element. This slack is taken up during installation, since the interference fit of the seal in standard O-ring grooves either stretches (piston seals) or compresses (rod seals) the element to seat the backups (See Figure 7). Parker T-seals are easy to assemble and may be inserted into their grooves manually without special assembly tools.

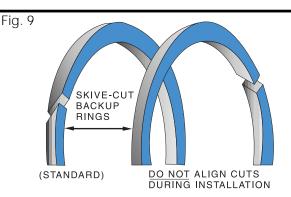




Seals

Introduction





The Parker T-seal back-up rings' ability to follow changing dynamic surfaces rapidly also makes them very effective wipers, keeping contaminants away from the sealing line (See Figure 8).

SPLIT vs. ENDLESS BACK-UPS

Standard devices in the Parker Aerospace T-Seals use split (skive-cut) back-up rings. Skive-cut rings are simple to install and allow harder and more extrusion-resistant materials to be used without sacrificing their ability to respond quickly to gap changes (See Figure 9). Endless (uncut) rings are normally made of PTFE, which is easily deformed for installation. An exception to this rather broad rule could be made in the case of assembled glands, where the back-up rings need not be excessively stretched or otherwise deformed during installation. In such cases, endless back-ups made of filled nylon or even harder compounds may be installed by direct insertion into an open-end groove, which is subsequently closed by a gland end-piece, which results in loss of positive actuation.

In most cases, Parker Aerospace T-seals may be inserted into their grooves manually without special assembly tools. Where endless nylon back-ups are used, however, it may be necessary to employ installation cones for piston seal back-ups and folding tools

for rod seal back-ups, provided small cross sections and thicknesses are being installed.

Special care must be taken when installing radiused back-ups. The radius of the back-up rings must be nested against the radiused surface of the seal element. Improper installation will compromise the sealing ability of the T-seal (See Figures 10 and 11).

FINITE ELEMENT ANALYSIS

Parker utilizes Finite Element Analysis (FEA) tools to simulate installation and operating conditions of your application. As shown in Figure 12 below, correct installation with proper nesting of back-up rings along the radiused corner of the elastomeric T-seal element concentrates the sealing force against the dynamic surface of the application.

Fig. 12

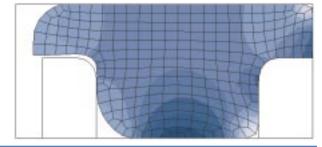


Fig. 10

WRONG!!

Incorrect installation of Aerospace T-Seal with radiused back-up rings.



Fig. 11

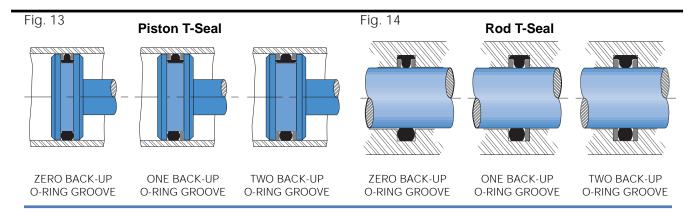
RIGHT!!

Correct installation of Aerospace T-Seal with radiused back-up rings.





Custom Designs



Parker's Aerospace T-seals are available to fit zero, one and two back-up width AS4716/MIL-G-5514F Oring grooves (See Figures 13 and 14). The standard groove width for Parker Aerospace T-Seals is the "zero back-up" O-ring groove design. One and two back-up widths are available. All standard Aerospace T-seal assemblies have two back-ups regardless of the gland width design.

When specifing Parker Aerospace T-seals for zero, one or two back-up groove widths, use the following part number designation:

Zero back-up O-ring groove:

N304TA0P1210110E

One back-up O-ring groove:

N304TA1P1210110E

Two back-up O-ring groove:

N304TA2P1210110E

See Page 7 of this catalog for complete part numbering instructions.

CUSTOM DESIGNS

Parker Applications Engineers can assist you in designing custom Aerospace T-seals for out-of-the ordinary applications. Such applications may include the design of dual staged back-ups for actuators, limited dynamics, or high pressure conditions. Custom designs may also be necessary where specialty compounds are warranted.

DUAL STAGED BACK-UPS

While applications which require dual material backups on each side of the seal are not a standard design, Parker Applications Engineers can assist you in designing Aerospace T-seals with dual-staged back-up rings to your specifications for applications warranting the same. Great care must be taken in installation of dual-staged back-ups so that the harder material back-up ring is installed adjacent to the extrusion gap, and the softer material back-up ring is installed against the seal body (See Figure 15).

Call the factory for special part numbering configurations on Parker Aerospace T-Seals with staged back-up rings.

PERFLUORINATED ELASTOMERS

Table 1. Perfluorinated Elastomers

Perfluorinated elastomers are used where better resistance to aggressive chemicals or extreme temperature resistance is needed (See Table 1).

Fig. 15

Base Elastomer Compound Temp Range °F Number Duro. Service HiFluor™ 3819 -15 to +400 Better resistance to 75 3534 90 -15 to +400 aggressive chemicals Parofluor™ 8545 75 -15 to +500/550 Extreme chemicals 8588 90 -15 to +500/550 Extreme temps



Parker Aerospace T-Seal Materials



Materials used for Parker Aerospace T-Seal assemblies are chosen for their abrasion resistance, fluid compatibility and temperature range, but the seal element and back-up rings have additional requirements unique to themselves. The seal element, for example, must also have high resistance to compression set, provide good adhesion for a lubricant film, high tear resistance, and long retention of its resilience. Back-up rings, in addition to the common requirements, must also have high shear strength,

sufficient hardness to bridge large gaps without extrusion, and either the ability to hold a lubricant film or have self-lubricating properties.

SEALING BODY ELASTOMER MATERIALS

Parker's Aerospace T-Seal series provides sealing bodies in four different standard elastomers (though literally dozens of alternate materials may be specified for special duty). These four elastomeric compounds include:

N304-75 NITRILE (Buna-N or NBR) is one of the most oil-resistant of the conventional seal polymers. It is widely used to seal petroleum-based hydraulic fluids (MIL-H-5606), water/oil emulsions, ethylene glycol fluids, and pneumatic systems where petroleum-based lubricants are added to the gas supply. The N304 compound meets MIL-P-25732, at -65°F to +275°F. Besides meeting all the general and special requirments of T-seal sealing elements, nitriles also remain resilient at -65°F, thereby minimizing leakage during cold start-ups.

E1267-80 ETHYLENE PROPYLENE (EPR or EPDM) is particularly useful in low density and high density phosphate ester fluids such as the Skydrol series, water-glycol fluids, long-term water service, and brake fluids from -65°F to +300°F. Do **not** use ethylene propylene with petroleum lubricants, solvents or other petroleum-based fluids, and especially avoid the practice of lubing the seals during installation with petroleum oils or greases.

V747-75 FLUOROCARBON (FKM, FPN) is a general purpose, 75 Shore A hardness fluorocarbon. Its excellent compression set resistance, high temperature and high pressure resistance make it excellent in sealing petroleum oils and fuels. The V747 compound meets MIL-R-83248. Temperature range is -15°F to +400°F.

V709-90 FLUOROCARBON (FKM, FPN) is a general purpose, 90 Shore A hardness fluorocarbon. Its

excellent compression set resistance, high temperature and high pressure resistance make it excellent in sealing petroleum oils and fuels. The V709 compound meets MIL-R-83248. Temperature range is -15°F to +400°F.

BACK-UP RING MATERIALS

As with sealing element compounds, Parker's Aerospace T-Seal series is available in four different standard back-up ring compounds. These materials are non-elastomeric, and are extremely resistant to extrusion.

POLYTETRAFLUOROETHYLENE (PTFE) per MIL-R-8791 is a self-lubricating polymer with the wide temperature range of -350°F to +400°F, and is able to tolerate any fluids found in aerospace service.* PTFE tends to cold flow, and is recommended for pressures up to 3000 psi with maximum diametral clearances of 0.0025 inch.

FILLED PTFE has improved extrusion resistance, wear and temperature ranges by the use of compounding fillers, which increases PTFE's hardness throughout its temperature range. Filled PTFE raises the maximum pressure recommendations (See Table 3)

NYLATRON (Polyamide Resin with Molybdenum Disulfide filler) is also self-lubricating, and also has the ability to hold a strongly adhering film of system lubricant on its surface. Nylatron is extremely tough and wear-resistant, and tolerates most petroleum-based fluids, phosphate ester hydraulic fluids, fuels, lubes, and solvents. It is more resistant to cold flow from -65°F to +250°F than PTFE, though PTFE can actually tolerate higher temperatures. Unlike PTFE, Nylatron does not interfere with the adhesion of lubricating films to the dynamic surfaces. Nylatron may be used to 5000 psi with diametral clearances of 0.0025 inch.

*Overall temperature range dependent on sealing body elastomer.



Materials

Sealing Body Elastomeric Materials:

Table 2. Standard Elastomeric Aerospace T-Seal Compounds

			Servio	e	
Compound Reference Number	Compound Type	Hardness (Shore A)	Eluide	Temp Range (°F)	Comments/Applications
N304	Nitrile	75	General purpose sealing, petroleum oils and fluids, cold water	-65° to +275°	MIL-R-25732 Mil-R-3065 Mil-Std -417 (Military aircraft petroleum-based hydraulic fluids, fuels, lubes)
1267	Ethylene Propylene	80	Phosphate ester- based hydraulic fluids (Skydrol)	-65° to +300°	NAS 1613
V747	Fluoro- carbon	75	Petroleum oils, di- ester base lubri- cants, silicone fluids and greases	-15° to +400°	MIL-R-83248
V709	Fluoro- carbon	90	Petroleum oils, di- ester base lubri- cants, silicone fluids and greases	-15° to +400°	MIL-R-83248

Back-up Ring Materials:

Table 3. Standard Back-up Ring Materials for Aerospace T-Seals

Compound Reference Number	Material	Color	Temp. Range (°F)*	Comments
0110	Virgin PTFE	White	-350° to +400°	Best for static applications requiring positive sealing. Mil-R-8791. Good in vacuum with low gas permeability. Low particulate generation. Excellent in cryogenics. Can be used in slow, infrequent dynamics. Recommended to 3000
0008	Carbon Fiber/PTFE	Brown	-250° to +550°	Excellent all-purpose material. Best for dynamic applications running on moderate to hard surfaces. High wear material with low abrasion. Recommended to 5000 psi.
0245	Graphite/PTFE	Dk. Gray	-250° to +550°	A dynamic material for softer surfaces and a static material for high temperatures. Good wear resistance with very low abrasion. Recommended to 4000 psi.
4655	Nylatron	Gray	-65° to +250°	Excellent chemical resistance. Recommended to 5000 psi on Dash Size # -006 to -149. Recommended to 10000 psi on Dash Size # -210 to -460.

^{*}Overall temperature range dependent on sealing body elastomer.

NOTICE:

Parker Packing Applications Engineers will be pleased to suggest special compounds for special applications where standard materials fall short of requirements. Many additional materials are already available as options to the standards, and still more may be qualified to unusual requirements.



Part Numbering

Part Numbering



Parker AerospaceT-Seals may be completely described and ordered by a seven-part composite Part Number (see below). Alpha-numeric codes define all standard seal dimensions, materials and configurations. Special materials or sizes, should be ordered only after consultation with Parker Packing's Applications Engineers, who will help you construct the proper Part Number. The materials and dimensions given in Tables 4 through 8 accommodate most aviation hydraulic and pneumatic requirements, but Parker has many

other materials, combinations of materials and special-size capability on request. For special design assistance and custom part numbering, photocopy pages 13 and 14 of this catalog, fill in the required information and fax or mail the same to Parker's Applications Engineers.

Parker Aerospace T-Seal Part Numbers may be constructed as follows:

EXAMPLE: A piston T-seal is required for a cylinder bore of 1.241 inch, to replace an O-ring in a zero back-up AS4716/MIL-G-5514F groove, operating in MIL-H-5605 fluid at 3000 psi and 200°F.

This Part Number would be written in these seven steps:

Elastomer Material	Aircraft T-Seal	Groove Width 0 B/U = 0 1 B/U = 1 2 B/U = 2	Piston / Rod "P" "R"	Dash Size AS4716/ Mil-G-5514F	Back-Up Ring Material	Endless or Split Backup "E" / "S"
N304	ТА	0	Р	121	0110	E

N304 TA, O, P 121 0110 E

Step 1: Select Sealing Body Designator for Piston or Rod Step 4:

Elastomeric Material. seal. (See Table 2, Page 6.)

Step 5: Dash Size Number AS4716/MIL-G-5514F. Step 2: Designator for Aerospace

T-Seal. (See Tables 4-8, Pages 8-12.)

Groove Width. Standard Step 3: Step 6: Select Back-up Ring Material.

> groove width is the zero back-(See Table 3, Page 6.)

up groove width.

(See Table 3, Page 6.) Step 7: Designator for Endless or Split

Back-up Ring.



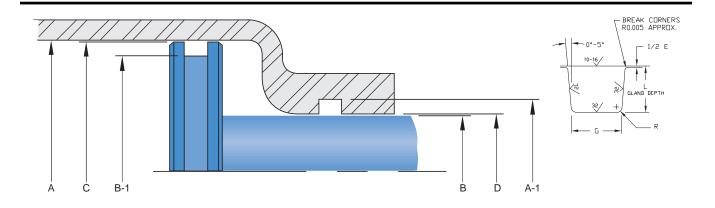


Table 4. Gland Dimensions AS4716 / Mil-G-5514F.

		P	iston			Rod					Groc	ve W	/idth
AS4716 Dash Size Number	Bore	Groove	Groove Dia. per Mil-G-	Piston	Max. Diametral Clearance	Rod	Groove	Groove Dia. per Mil-G-	Bore	Max. Diametral Clearance	(Back-u	p Groove	e Width
	Dia.	Dia. B-1	5514F B-1	OD C	_≝ŏ E	Dia. B	Dia. A-1	5514F A-1	Dia.	E E	0	1	2
	+.001	+.000	+.000	+.001	_	+.000	+.001 000	+.001	+.000		+.005	+.010 000	+.010 000
-006 -007 -008 -009	.235 .266 .297 .329	.123 .154 .189 .220	.123 .154 .185 .217	.233 .264 .295 .327	.004	.123 .154 .185 .217	.235 .266 .294 .327	.235 .266 .297 .329	.125 .156 .187 .219	.004	.098*	.154 [†]	.210 [‡]
-010 -011 -012	.360 .422 .485	.250 .312 .375	.248 .310 .373	.358 .420 .483	+	.248 .310 .373	.359 .421 .484	.360 .422 .485	.250 .312 .375	↓	1	1	1
-013 -014 -015 -016 -017 -018 -019 -020 -021 -022 -023 -024 -025 -026	+.002 000 .550 .613 .675 .738 .800 .863 .925 .991 1.053 1.116 1.178 1.241 1.303 1.366	.441 .504 .566 .629 .691 .753 .815 .881 .943 1.006 1.068 1.131 1.193 1.256	+.000 002 .438 .501 .563 .626 .688 .751 .813 .879 .941 1.004 1.066 1.129 1.191 1.254	+.002 000 .548 .611 .673 .736 .798 .861 .923 .989 1.051 1.114 1.176 1.239 1.301 1.384	.005	.435 .498 .560 .623 .685 .748 .810 .873 .935 .998 1.060 1.123 1.185	.545 .608 .670 .733 .795 .858 .920 .983 1.045 1.108 1.170 1.233 1.295 1.358	+.002 000 .547 .610 .672 .735 .797 .860 .922 .985 1.047 1.110 1.172 1.235 1.297 1.360	+.000 002 .437 .500 .562 .625 .687 .750 .812 .875 .937 1.000 1.062 1.125 1.187 1.250	.005	.094	.150 [†]	.207

*-006 to -009 †-006 to -028 0 Groove Width

.094 .149

‡-006 to -009

1 Groove Width 2 Groove Width .149 .207



Table 5. Gland Dimensi	ons AS4716 / Mil-G-5514F.
------------------------	---------------------------

		P	iston				1	Rod			Groove Width			
AS4716 Dash Size Number	Bore Dia.	Groove Dia.	Groove Dia. per Mil-G- 5514F	Piston OD	Max. Diametral Clearance	Rod Dia.	Groove Dia.	Groove Dia. per Mil-G- 5514F	Bore Dia.	Max. Diametral Clearance		p Groove		
	Α	B-1	B-1	С	Е	В	A-1	A-1	D	Е	0	1	2	
	+.002	+.000 002	+.000 002	+.002 000		+.000 002	+.002	+.002	+.000		+.010 000	+.010 000	+.010	
-110 -111 -112 -113 -114 -115 -116 -117 -118 -119 -120 -121 -122 -123 -124 -125 -126* -126* -130 -131 -132‡ -133 -134 -135 -136 -137 -138 -139 -140 -141 -142 -143 -144 -145 -146 -147 -148 -149	.550 .613 .675 .738 .800 .863 .925 .991 1.053 1.116 1.178 1.241 1.303 1.366 1.428 1.491 1.553 1.616 1.678 1.741 1.805 1.867 1.930 1.992 2.055 2.118 2.180 2.243 2.305 2.368 2.430 2.493 2.555 2.618 2.680 2.743 2.805 2.868 2.930 2.993	.379 .441 .502 .565 .627 .689 .751 .817 .879 .942 1.003 1.066 1.128 1.191 1.253 1.316 1.378 1.441 1.503 1.566 1.631 1.693 1.756 1.818 1.881 1.944 2.006 2.069 2.131 2.194 2.256 2.319 2.381 2.444 2.506 2.569 2.631 2.694 2.756 2.819	.372 .435 .497 .560 .622 .685 .747 .813 .875 .938 1.000 1.063 1.125 1.188 1.250 1.313 1.375 1.438 1.500 1.563 1.627 1.689 1.752 1.814 1.877 1.940 2.002 2.065 2.127 2.190 2.252 2.315 2.377 2.440 2.502 2.565 2.627 2.690 2.752 2.815	.548 .611 .673 .736 .798 .861 .923 .989 1.051 1.114 1.176 1.239 1.301 1.364 1.426 1.489 1.551 1.614 1.676 1.739 1.802 1.864 1.927 1.989 2.052 2.115 2.177 2.240 2.302 2.365 2.427 2.490 2.552 2.615 2.677 2.740 2.802 2.865 2.927 2.990	.005	.373 .435 .498 .560 .623 .685 .748 .810 .873 .935 .998 1.060 1.123 1.185 1.248 1.310 1.373 1.435 1.685 1.748 1.560 1.623 1.685 1.748 1.810 1.873 1.936 1.998 2.061 2.123 2.186 2.248 2.311 2.373 2.436 2.498 2.561 2.623 2.686 2.748 2.811	.546 .609 .672 .734 .797 .859 .923 .985 1.048 1.110 1.173 1.235 1.298 1.360 1.423 1.485 1.548 1.610 1.673 1.735 1.798 1.860 1.923 1.984 2.047 2.110 2.172 2.235 2.297 2.360 2.422 2.485 2.547 2.610 2.672 2.735 2.797 2.860 2.922 2.985	.551 .613 .676 .738 .801 .863 .926 .988 1.051 1.113 1.176 1.238 1.301 1.363 1.426 1.488 1.551 1.613 1.676 1.738 1.801 1.863 1.926 1.988 2.051 2.114 2.176 2.239 2.301 2.364 2.426 2.426 2.489 2.551 2.614 2.676 2.739 2.801 2.864 2.926 2.928	.375 .437 .500 .562 .625 .687 .750 .812 .875 .937 1.000 1.062 1.125 1.187 1.250 1.312 1.375 1.437 1.500 1.562 1.625 1.687 1.750 1.813 1.876 1.939 2.001 2.064 2.126 2.189 2.251 2.376 2.439 2.501 2.564 2.626 2.689 2.751 2.814	.005	.141	.183	.245	

*-126 .005 Rod Max. Diametral Clearance †-129 .006 .007

Piston Max. Diametral Clearance Rod Max. Diametral Clearance ‡-132



Table 6. Gland Dimensions AS4716 / Mil-G-5514F.

		Р	iston					Rod			Groove Width			
AS4716 Dash Size Number	Bore Dia.	Groove Dia.	Groove Dia. per Mil-G- 5514F	Piston OD	Max. Diametral Clearance	Rod Dia.	Groove Dia.	Groove Dia. per Mil-G- 5514F	Bore Dia.	Max. Diametral Clearance		p Groove G		
	Α	B-1	B-1	С	Е	В	A-1	A-1	D	Е	0	1	2	
	+.002 000	+.000 002	+.000 002	+.002 000		+.000 002	+.002 000	+.002 000	+.000 002		+.010 000	+.010 000	+.010 000	
-210 -211 -212 -213 -214 -215 -216 -217 -218 -219 -220 -221 -222 -223 -224 -225 -226 -227* -228 -229 -230 -231 -232 -233 -234 -235 -236 -237 -238 -239 -240 -241 -242 -243 -244 -245 -246 -247	.991 1.053 1.116 1.178 1.241 1.303 1.366 1.428 1.491 1.553 1.616 1.678 1.741 1.867 1.992 2.118 2.243 2.368 2.493 2.618 2.743 2.868 2.993 3.118 3.243 3.368 3.493 3.618 3.743 3.868 3.993 4.118 4.243 4.368 4.493 4.368 4.493 4.618 4.743 4.868	.750 .812 .874 .936 .999 1.064 1.124 1.186 1.249 1.311 1.374 1.436 1.499 1.625 1.750 1.876 2.001 2.126 2.251 2.376 2.501 2.626 2.751 2.876 3.001 3.126 3.251 3.376 3.501 3.501 3.626 3.751 3.876 4.001 4.126 4.251 4.376 4.501 4.626	.748 .810 .873 .935 .998 1.060 1.123 1.185 1.248 1.310 1.373 1.435 1.498 1.624 1.749 1.875 2.000 2.125 2.250 2.375 2.500 2.625 2.750 2.875 3.000 3.125 3.250 3.375 3.500 3.625 3.750 3.875 4.000 4.125 4.250 4.375 4.500 4.625	.989 1.051 1.114 1.176 1.239 1.301 1.364 1.426 1.489 1.551 1.614 1.676 1.739 1.864 1.989 2.115 2.240 2.365 2.490 2.615 2.740 2.865 2.990 3.115 3.240 3.365 3.740 3.365 3.740 3.865 3.740 3.865 3.740 4.15 4.240 4.365 4.289 4.115 4.240 4.365 4.289 4.115	.005	.748 .810 .873 .935 .998 1.060 1.123 1.185 1.248 1.310 1.373 1.435 1.498 1.623 1.748 1.873 1.998 2.123 2.248 2.373 2.248 2.373 2.248 2.373 2.498 2.623 2.748 2.873 2.997 3.122 3.247 3.372 3.497 3.622 3.747 3.872 3.997 4.122 4.247 4.372 4.497 4.622	.989 1.051 1.115 1.177 1.240 1.302 1.365 1.427 1.490 1.552 1.615 1.677 1.740 1.865 1.990 2.115 2.240 2.365 2.490 2.615 2.740 2.865 2.990 3.115 3.239 3.364 3.489 3.614 3.739 3.864 3.739 3.864 3.739 4.4489 4.614 4.739 4.864	.991 1.053 1.116 1.178 1.241 1.303 1.366 1.428 1.491 1.553 1.616 1.678 1.741 1.866 1.991 2.116 2.241 2.366 2.491 2.616 2.741 2.866 2.991 3.116 3.240 3.365 3.490 3.615 3.740 3.865 3.740 4.365 4.490 4.115 4.240 4.365 4.490 4.615 4.740 4.865	.750 .812 .875 .937 1.000 1.062 1.125 1.187 1.250 1.312 1.375 1.437 1.500 1.625 1.750 1.876 2.001 2.126 2.251 2.376 2.501 2.626 2.751 2.876 3.000 3.125 3.250 3.375 3.500 3.625 3.750 3.875 4.000 4.125 4.250 4.250 4.375 4.501 4.626	.005	.188	.235	.304	

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Piston Max. Diametral Clearance .007



Table 7. Gland Dimensions AS4716 / Mil-G-5514F.

		P	iston					Rod			Gro	ove W	/idth
AS4716 Dash Size Number	Bore Dia.	Groove Dia.	Groove Dia. per Mil-G- 5514F	Piston OD	Max. Diametral Clearance	Rod Dia.	Groove Dia.	Groove Dia. per Mil-G- 5514F	Bore Dia.	Max. Diametral Clearance	`	p Groove	
	Α	B-1	B-1	С	E	В	A-1	A-1	D	E	0	1	2
	+.002 000	+.000 002	+.000 002	+.002 000		+.000 002	+.002 000	+.002	+.000 002		+.010	+.010 000	+.010 000
-325 -326 -327 -328 -329 * -330 -331 -332 -333 -334 -335 -336 -337 -338 -340 -341 -342 -343 -344 -345 -346 -347 -348 -349	1.867 1.992 2.118 2.243 2.368 2.493 2.618 2.743 2.868 2.993 3.118 3.243 3.368 3.493 3.618 3.743 3.868 3.993 4.118 4.243 4.368 4.493 4.618 4.743 4.868	1.495 1.620 1.746 1.871 1.996 2.121 2.246 2.371 2.496 2.621 2.746 2.871 2.996 3.121 3.246 3.371 3.496 3.621 3.746 3.871 3.996 4.121 4.246 4.371 4.496	1.495 1.620 1.746 1.871 1.996 2.121 2.246 2.371 2.496 2.621 2.746 2.871 2.996 3.121 3.246 3.371 3.496 3.621 3.746 3.871 3.996 4.121 4.246 4.371 4.496	1.864 1.989 2.115 2.240 2.365 2.490 2.615 2.740 2.865 2.990 3.115 3.240 3.365 3.490 3.615 3.740 3.865 3.990 4.115 4.240 4.365 4.489 4.614 4.739 4.864	.006	1.498 1.623 1.748 1.873 1.998 2.123 2.248 2.373 2.498 2.623 2.748 2.873 2.997 3.122 3.247 3.372 3.497 3.622 3.747 3.872 3.997 4.122 4.247 4.372 4.497	1.870 1.995 2.120 2.245 2.370 2.495 2.620 2.745 2.870 2.995 3.120 3.245 3.369 3.494 3.619 3.744 3.869 4.119 4.244 4.369 4.494 4.619 4.744 4.869	1.870 1.995 2.120 2.245 2.370 2.495 2.620 2.745 2.870 2.995 3.120 3.245 3.369 3.494 3.619 3.744 3.869 3.994 4.119 4.244 4.369 4.494 4.619 4.744 4.869	1.500 1.625 1.750 1.876 2.001 2.126 2.251 2.376 2.501 2.626 2.751 2.876 3.000 3.125 3.250 3.375 3.500 3.625 3.750 3.875 4.000 4.125 4.250 4.375 4.500	.006	.281	.334	.424

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Table 8. Gland Dimensions AS4716 / Mil-G-5514F.

		P	iston					Rod			Groove Width				
AS4716 Dash Size Number	Bore Dia.	Groove Dia.	Groove Dia. per Mil-G- 5514F	Piston OD	Max. Diametral Clearance	Rod Dia.	Groove Dia.	Groove Dia. per Mil-G- 5514F	Bore Dia.	Max. Diametral Clearance		p Groove			
	Α	B-1	B-1	С	Е	В	A-1	A-1	D	Е	0	1	2		
	+.003 000	+.000 003	+.000	+.003 000	+.003 000	+.000 003	+.003 000	+.003 000	+.000		+.010 000	+.010 000	+.010		
-425	4.974	4.497	4.497	4.970 5.095	A	4.497	4.974	4.974	4.501	A					
-426 -427	5.099 5.224	4.622 4.747	4.622 4.747	5.095		4.622	5.099	5.099	4.626 4.751						
-427 -428	5.349	4.747	4.747	5.220		4.747 4.872	5.224 5.349	5.224 5.349	4.731						
-420 -429	5.474	4.872	4.072	5.470		4.872	5.474	5.474	5.001						
-429	5.599	5.122	5.122	5.595		5.122	5.599	5.474	5.126						
-430	5.724	5.122	5.122	5.720		5.122	5.724	5.724	5.120	.009					
-432	5.849	5.372	5.372	5.845		5.372	5.849	5.724	5.376						
-433	5.974	5.497	5.497	5.970		5.372	5.974	5.974	5.501						
-434	6.099	5.622	5.622	6.095		5.622	6.099	6.099	5.626						
-435	6.224	5.747	5.747	6.220	'	5.747	6.224	6.224	5.751						
-436	6.349	5.872	5.872	6.345	.009	5.872	6.349	6.349	5.876						
-437	6.474	5.997	5.997	6.470		5.997	6.474	6.474	6.001	▼					
-438	6.724	6.247	6.247	6.720		6.247	6.724	6.724	6.251						
-439	6.974	6.497	6.497	6.970		6.497	6.974	6.974	6.501						
-440	7.224	6.747	6.747	7.220		6.747	7.224	7.224	6.751	♠					
-441	7.474	6.997	6.997	7.470		6.997	7.474	7.474	7.001						
-442	7.724	7.247	7.247	7.720		7.247	7.724	7.724	7.251		'	1			
-443	7.974	7.497	7.497	7.970		7.497	7.974	7.974	7.501		.375	.475	.579		
-444	8.224	7.747	7.747	8.220	↓	7.747	8.224	8.224	7.751		١.	l ,			
-445*	8.474	7.997	7.997	8.470	▼	7.997	8.474	8.474	8.001						
-446	8.974	8.497	8.497	8.970	.010	8.497	8.974	8.974	8.501						
	+.004 000					01177	+.004	+.004							
-447	9.474	8.997	8.997	9.470	l ▲	8.997	9.474	9.474	9.001						
-448	9.974	9.497	9.497	9.970	lt	9.497	9.974	9.974	9.501						
-449	10.474	9.997		10.470		9.997		10.474		010					
-450		10.497				10.497		10.974		.010					
-451	11.474	10.997		11.470		10.997	11.474	11.474							
-452	11.974	11.497		11.970		11.497	11.974	11.974							
-453	12.474	11.997		12.470	011	11.997	12.474	12.474							
-454	12.974	12.497	12.497	12.970	.011	12.497	12.974	12.974							
-455	13.474	12.997		13.470	ı	12.997	13.474	13.474							
-456	13.974	13.497		13.970		13.497	13.974	13.974							
-457	14.474	13.997	13.997	14.470		13.997	14.474	14.474							
-458	14.974	14.497	14.497	14.970		14.497	14.974	14.974							
-459	15.474	14.997		15.470		14.997	15.474	15.474							
-460		15.497		15.970	↓	15.497	15.974	15.974		\	\ \	♦	₩		
Mil-G-551					•										

*-445

Piston Max. Diametral Clearance .010



Data Design Request Form

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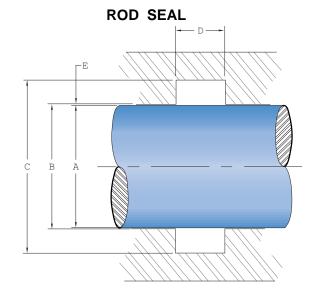
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E-MAIL			E-MAIL			
COMPONENTS			—— PRO	BLEM:		
EXISTING SEAL/PAR	T NUMBER:	Dı	N/A			
PRICE \$ @	PCS_USAGE/Y	EAR:				
TARGET \$ @ _	PCS QUOTE (QTY:				
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	SEAL PROBLEM (Expla					
PROTO QTY:	DATE PROTO REG	Q'D:	CUS	TOMER P/N:		
SPECIAL INSPECTION	N REQUIREMENTS? YE	ES 🗖 NO	SPECIAL	PACKAGING RE	EQUIREMENTS?	□YES □NO
PRODUCT TYPE						
□ ROD/SHAFT	PISTON		INTERNAL FA	CE	■ EXTERNAL I	FACE
OPERATING	UNIT					
PARAMETERS	(CIRCLE ONE)	MINIMUM C	PERATING	MAXIMUM	MOTION	
TEMPERATURE	°K °F °C		21 210 11110		STATIC	
PRESSURE	PSI BAR MPA				RECIPRO	CATINIC
	-					JATING
STROKE LENGTH	INCH MM				ROTARY	
CYCLE RATE	/MIN. /HR HZ					ORY
ROTATION	DEG. RAD.					
RPM					PRESSURE D	DIRECTION
VELOCITY/					UNIDIREC	TIONAL
SURFACE SPEED	FT/MIN. MM/MIN.				☐ BI-DIRECT	TONAL
VACUUM	IN. HG TORR					
DIRECTION OF ROTA	ATION CLOC	CKWISE COL	JNTER CLOCK	WISE 🗖 BI-DI	RECTIONAL	
MEDIA TO BE SEA	LED					

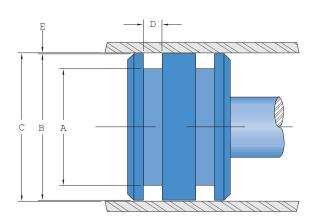


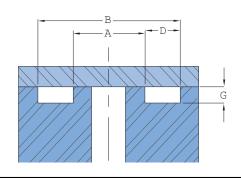
Data Design Request Form

HARDWARE









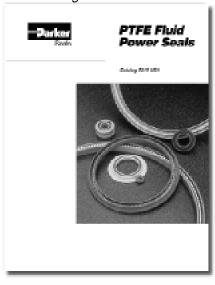
HARDWARE DRAWINGS INCLUDED WITH DAT	TA REQUEST	□YES □NO	STEP C	CUT YES NO
B DIAMETER MIN C DIAMETER MIN D GROOVE WIDTH MIN E RADIAL CLEARANCE MIN F ROD/PISTON STEP HEIGHT MIN G FACE SEAL GROOVE DEPTH MIN	MAX	HARDNESS HARDNESS GLAND TY SPLIT SOLID	FINISH _ FINISH PE □ OPEN	MAT'L MAT'L MAT'L MAT'L MAT'L METRIC YES NO
SIDE LOAD (LBS NEWTONS):			_	_
AS4716/MIL-G-5514F DASH #	BACK-UP WIDTH:	∐ ZERO L	ONE	□TWO
PERFORMANCE REQUIREMENTS (CIRCLE ONE) FRICTION LBS OZ EXPECTED LIFE: CYC HRS MAX LEAKAGE: DROPS CC/MIN	YRS	AKOUT 🗖 DYNAM		
MOST CRITICAL ASPECT (check only one)	SEAL LIFE PR	ICE LEAKAGE [FRICTION C	OTHER (EXPLAIN)
CONTAMINATION:				





SPRING-ENERGIZED RUBBER U-CUPS

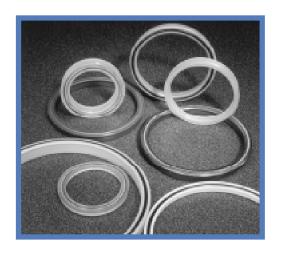
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